

WHAT IS CLAIMED IS:

1 1. An illuminating-reflector system for transmitting a frequency band in
2 an dispersed beam and a substantially collimated beam, the system comprising:
3 a secondary reflector configured to transmit a first portion of the frequency
4 band to form the dispersed beam and to reflect a second portion of the frequency band; and
5 a primary reflector configured to receive the second portion of the frequency
6 band reflected from the secondary reflector and to reflect the second portion of the frequency
7 band to form the substantially collimated beam.

1 2. The system of claim 1, further comprising a dispersive lens configured
2 to receive the frequency band and transmit the frequency band to the secondary reflector in
3 another dispersed beam.

1 3. The system of claim 2, wherein the primary reflector includes an
2 aperture formed therein to pass the frequency band from the dispersive lens to the secondary
3 reflector.

1 4. The system of claim 2, wherein the dispersive lens is configured to
2 receive the frequency band from a beam waveguide.

1 5. The system of claim 1, wherein the first portion includes about twenty
2 percent or less of the power of the frequency band.

1 6. The system of claim 1, wherein the second portion includes about
2 eighty or more of the power of the frequency band.

1 7. The system of claim 1, wherein the frequency band includes a V-band
2 or a W-band.

1 8. The system of claim 7, wherein the V-band and the W-band
2 respectively include a Military Satellite Communications V-band and a Military Satellite
3 Communications W-band.

1 9. The system of claim 1, wherein the primary reflector has a diameter
2 greater than or equal to about six feet and less than or equal to about eight feet.

1 10. The system of claim 1, wherein the secondary reflector has a diameter
2 of greater than or equal to about 8 inches.

1 11. The system of claim 10, wherein the secondary reflector has a diameter
2 of about 12 inches.

1 12. The system of claim 1, wherein the secondary reflector is a compound
2 optical element.

1 13. The system of claim 1, wherein a gain of the primary reflector is
2 greater than or equal to about 50 dBi.

1 14. The system of claim 1, wherein a gain of the primary reflector is about
2 59.5 dBi.

1 15. The system of claim 1, wherein a gain of the secondary reflector is less
2 than or equal to about -33 dBi below the primary beam.

1 16. The system of claim 1, further comprising control electronics disposed
2 in a satellite bus and configured to control a transmission direction of the dispersed beam and
3 the substantially collimated beam.

1 17. The system of claim 1, wherein the dispersed beam is configured to be
2 acquired by a satellite for initial acquisition and automatic tracking of the system.

1 18. A satellite for cross-link communications with at least one other
2 satellite, the satellite comprising:
3 an illuminating reflector configured to transmit a first portion of a frequency
4 band in a collimated beam and a second portion of a frequency band in an dispersed beam.

1 19. The satellite of claim 18, wherein the dispersed beam is a low-gain
2 beam.

1 20. The satellite of claim 18, wherein the collimated beam is a high-gain
2 beam.

1 21. The satellite of claim 18, wherein the illuminating reflector includes:

2 a secondary reflector configured to transmit the first portion of the
3 frequency band to form the dispersed beam and to reflect a second portion of the
4 frequency band; and

5 a primary reflector configured to receive the second portion of the
6 frequency band reflected from the secondary reflector and to reflect the second
7 portion of the frequency band to form the substantially collimated beam.

1 22. The satellite of claim 18, wherein the dispersed beam is configured to
2 be acquired by another satellite for initial acquisition and automatic tracking of the first-
3 mentioned satellite.

1 23. The satellite of claim 18, further comprising a dispersive lens
2 configured to receive the frequency band from a beam waveguide and transmit the frequency
3 band to the secondary reflector.

1 24. The satellite of claim 23, wherein the primary reflector includes an
2 aperture formed therein to pass the frequency band transmitted from the dispersive lens to the
3 secondary reflector.

1 25. The satellite of claim 23, wherein the dispersive lens configured to
2 receive the frequency band from a beam waveguide.

1 26. The satellite of claim 23, wherein the first portion includes about five
2 percent or less of the power of the frequency band transmitted from the dispersive lens.

1 27. The satellite of claim 23, wherein the second portion includes about
2 ninety-five percent or more of the power of the frequency band transmitted from the
3 dispersive lens.

1 28. The satellite of claim 18, wherein the frequency band includes a W-
2 band.

1 29. The satellite of claim 28, wherein the W-band includes a Military
2 Satellite Communications W-band.

1 30. The satellite of claim 18, wherein the primary reflector has a diameter
2 greater than or equal to about six feet and less than or equal to about eight feet.

1 31. The satellite of claim 18, wherein the secondary reflector has a
2 diameter greater than or equal to about eight inches.

1 32. The satellite of claim 18, wherein a gain of the primary reflector is
2 greater than or equal to about 59 dBi.

1 33. The satellite of claim 32, wherein the gain of the primary reflector is
2 about 59.5 dBi.

1 34. The satellite of claim 18, wherein a gain of the secondary reflector is
2 less than or equal to about -33 dBi below the primary beam.

1 35. The satellite of claim 18, further comprising a satellite bus operatively
2 coupled to the illuminating reflector.

1 36. The satellite of claim 35, further comprising control electronics
2 disposed in the satellite bus and configured to slew the illuminating reflector.

1 37. A satellite communication method for cross-linked communication
2 between satellites, the method comprising:

3 at a first satellite:

4 transmitting in an dispersed beam a first portion of a frequency band
5 through a secondary reflector, wherein the secondary reflector is configured to form a
6 portion of an illuminating reflector;

7 reflecting a second portion of the frequency band from the secondary
8 reflector;

9 receiving at a primary reflector the second portion of the frequency
10 band reflected from the secondary reflector, wherein the primary reflector is
11 configured to form another portion of the illuminating reflector; and

12 reflecting at the primary reflector the second portion of the frequency
13 band to form a substantially collimated beam.

1 38. The method of claim 37, wherein the primary reflector and the
2 secondary reflector form a Cassegrain reflector.

1 39. The method of claim 37, further comprising:

2 at a second satellite:
3 acquiring the dispersed beam; and
4 tracking the dispersed beam to acquire the collimated beam.

1 40. The method of claim 39, further comprising transmitting a beacon
2 signal from the second satellite to the first satellite to indicate acquisition of the collimated
3 beam.

1 41. The method of claim 40, further comprising modulating the collimated
2 beam for communications in response to receiving the beacon signal.

1 -- 42. The method of claim 39, wherein the frequency band is un-modulated
2 prior to acquisition of the collimated beam by the second satellite.